## AGGA-4: core device for GNSS space receiver of this decade



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#### **Precise Orbit Determination (POD)**





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#### **Requirements in POD**



Key issues impose different GNSS receiver architectures and operational approach

- data timeliness (real-time OB and / or post-processing OG)
- robustness : high number of observations
- accuracy

Mission	Real Time	Non RT	Slow Time Critical	Non Time Critical		
	(RT)	(1-3h)	STC, (1-2 days)	(1 month)		
GOCE		< 50 cm rms	< 10 cm rms	< 2cm rms		
(launch: March 2009)		(requirement)	(ACHIEVED ~ 4 cm)	(ACHIEVED)		
Swarm		< 10 cm rms				
Sentinel-1 (SAR interferometry)	10 m. 3σ xyz	5 cm rms xyz				
Sentinel-3	3 m. rms	8 cm rms	3 cm rms	2 cm rms		
(Altimetry)	(radial)	(radial)	(radial)	(radial)		
MetOp-GRAS (Occultations) (launch: 2006)		0.1 mm/s (velocity. along) ACHIEVED				

## **Radio Occultation (RO)**





While a GNSS satellite 'sets' or 'rises' behind the horizon:

- > Additional bending of the GNSS signal's ray path due to refraction in the atmosphere
- The GNSS receiver measures the excess Doppler shift
  - $\Rightarrow$  key measurement is **CARRIER PHASE**
- derive vertical profiles (Temperature, Pressure, Humidty)

Performance is driven by very good clocks, open loop processing, high antenna gain ean Space Agency

#### **Future GNSS receiver architecture**







Attempt to make it as modular as possible (reproducibility & re-use) Difference POD and RO could be software and antenna

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## Baseband GNSS processor developed under ESA guidance and contracts



AGGA = Advanced GPS / Galileo ASIC

AGGA-2: [T7905E component] manufactured by Atmel in the year 2000

- Targeted for EO applications: POD, Radio Occultation (RO), attitude determination.
- Used in many missions:

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- ESA: e.g. MetOp-Gras a/b/c for RO, GOCE, Sentinels 1/2/3, Swarm, EarthCARE, etc.
- Non-ESA: e.g. ROSA in Oceansat Radarsat-2, Cosmo-Skymed, ...

Reasons to go for a new generation of devices

- new scientific requirements & experience from current instruments like MetOp GRAS
- new enhanced GNSS signals (GPS / Galileo / Compass / Glonass)
- Advances in space ASIC technology allowing more on-chip integration

#### AGGA-4 : Next generation with more functionality



#### In yellow the GNSS core

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#### **AGGA-4 overall architecture**



Legend **GNSS** core Ext. interfaces AGGA-4 LEON GPIO GPIO I/F Int. interface config On-chip modules TIMERs SPI I/F SPI Gaisler FPU Watchdog Trace buffer LEON2FT CIC Debug Debug IU UART / SpaceWire Support comm. write Status I-Cache D-Cache link Unit protect Arbiter/ PIC AHB AHB AHB AHB AHB Decoder SRAM AHB, А APB AHB PROM Mem H Ю В Ctrl **SDRAM** APB AHB | DMA AHB **MIL-Bus SpaceWire UARTs** FFT CRC 1553 **GNSS** core & GIC ∠<sub>36</sub> European Space Agency **\* \*** ¥ ¥ ★ ... GNSS Signal I/F Navitec - 2010 // 08-Dec-2010 SpaceWire I/F MIL-Bus I/F UART I/F 8

## AGGA-4 vs AGGA-2



	Feature	AGGA-4	AGGA-2		
	# of channels	<b>36</b> Single Freq. or 18 Dual Freq (target)	12 SF or 4 DF		
G N S S C H A N N E L S	Compatible signals	Galileo Open Service: E1bc, E5a, E5b Modernized GPS: L1 C/A, L1C, L2C, L5 Existing FDMA Glonass Potentially: Beidou, modernized Glonass	GPS L1 C/A Codeless L1/L2 Existing FDMA Glonass		
	Code Generators	(2 code generators per channel for Pilot and Data) Primary: LFSR and memory based Secondary codes and BOC(m,n) subcarriers	1 code generator per channel Fixed LFSR for certain primary codes only No secondary code and no BOC.		
	Correlators per channel	<b>5</b> complex (I/Q) with <b>EE</b> , E, P, L, <b>LL</b> (E=Early ; P=Punctual) and autonomous NAV data bit collection in HW	3 complex (I/Q), with E, P, L (L=Late) NAV data bit collection requires software interaction		
	Codeless P(Y) code	No	Yes ( 4 P-code units) – ESA patent		
	Channel Slaving	Hardware and software slaving	Hardware slaving		
	Aiding Unit per channel	Yes: Code and Carrier aiding	No. Done in software		
	Observables	16 Integration Epoch (IE) observables - DMA capable 5 Measurement Epochs (ME) observables - DMA capable	6 IE observables (no DMA - interrupt based) 2 ME observables (no DMA - interrupt based)		
	Common to all channels	Antenna Switch Controller (ASC) Time Base Generator (TBG)	ASC TBG		
	MICRO-PROCESSOR	LEON-2 FT on-chip with IEEE-754 compl. GRFPU Float.Point)	Off-chip (typically ERC-32, ADSP 21020)		
	INPUT FORMAT	<b>3 bit</b> (0.17 dB losses) (I/Q, real sampling and <b>interface for IF. ~ 250 MHz</b> )	2 bit (0.55 dB losses) (I/Q and real sampling)		
	CRC MODULE	Check Redundancy Code in hardware On-chip	No		
FFT MODULE		FFT in hardware on-chip	No		
INTERFACES		Two DMA capable UART, Mil-Std-1553, 4 SpaceWire SE, SPI I/F, DSU, S-GPO, 32 GPIO, SRAM I/F	Microprocessor I/F, Interrupt controller and I/O ports		
BEAMFORMING		Yes (2 Digital Beam Forming)	No		
TECHNOLOGY		0.18 Micron from ATMEL, 352 pins GNSS clock up to 50 MHz (target) – LEON clock target 80 MHz	0.5 micron from ATMEL, 160 pins GNSS clock up to 30 MHz		

#### AGGA-4 GNSS Core





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### **AGGA-4** Channel matrix



- \* 36 single-frequency double-code
- \* Very flexible primary code generator units:
  - a LFSR to generate very long codes
    - (e.g. 767,250 chips in L2CL)
  - memory-based codes
    - (e.g. for Galileo E1b and E1c).



- \* Support of Binary Offset Carrier BOC(m,n) and
  - secondary codes required in modernized GPS and new Galileo signals.
- \* <u>5 complex (I/Q) code correlators</u>, to allow the EE, E, Punctual, L, LL required for the processing of BOC signals.
- \* <u>hardware Aiding Unit</u>, allowing autonomous CODE and CARRIER aiding in order to compensate for the 'predictable' Doppler rate (Hz/s) caused by high orbit dynamics<sub>European Space Agency</sub>

#### Signals processed with AGGA-4



- Relying on Public signals (no PRS, SoL, ... )
- The double code generator allows to process the two component signals in one channel
- High flexibility => also compatible with GLONASS and Beidou (as known today)

Band	Freq. (MHz)	Compo nent	Code Rate (Mcps)	Primary code length (chips)	Secondary code length (chips)	Symbol/ Data Rate (sps / (bps)	Replicas in AGGA-4	LFSR/ Memory (config. AGGA4)	AGGA4 Nb. Channels
E1	1575.42	E1 B	1.023	4,092	No	250/125	BOC(1,1)	Memory	1 SF
		E1 C	1.023	4,092	25	Pilot	BOC(1,1)	Memory	(Sing. Freq.)
E5a	1176.45	E5a-I (E5b-I)	10.23 (idem)	10,230 (idem)	20 (4)	50/25 (250/125)	BPSK(10) (idem)	LFSR (idem)	1 SF
(E5b)	(1207.14)	E5a-Q (E5b-Q)	10.23 (idem)	10,230 (idem)	100 (idem)	Pilot	BPSK(10) (idem)	Memory (idem)	(idem)
L1c	1575.42	L1Cd	1.023	10,230	No	100/50	BOC(1,1)	Memory	1 SF
		L1Cp	1.023	10,230	1800	Pilot	BOC(1,1)	Memory	1 SF
L1	1575.42	L1 C/A	1.023	1,023	No	50	BPSK(1)	LFSR	1 SF
L2C	1227.6	L2CM	10.23	10,230	No	50/25	BPSK(0.5)	Memory	1 SF
		L2CL	10.23	767,250	No	Pilot	BPSK(0.5)	LFSR	
L5	1176.45	L5-I	10.23	10,230	10	100/50	BPSK(10)	LFSR	1 SF
		L5-Q	10.23	10,230	20	Pilot	BPSK(10)	Memory	

#### **AGGA-4** schedule



- AGGA-4 is under development by **<u>Astrium GmbH</u>** under ESA guidance and Contracts.
- Extensive validation (acquisition & tracking) with:
  - FPGA version (same as ASIC but with only 4 GNSS channels)
  - Block testing and use of E2E testing with Spirent simulator at ESTEC
  - by Ruag Space Austria in August 2010
- Deimos Engenharia also contributing to FPGA validation
- ASIC components by Atmel in ATC18RHA 0.18 μm process (MQFP package with 352 pins).
  - available for the whole European space industry (equal basis).
- sis). Multi-interface AGGA-4 FPGA testing board

- ASIC prototypes by 4Q-2011.





AGGA-4 FPGA testing at ESTEC in Aug. 2010 with Spirent simulator ec - 2010 // 08-Dec-2010

#### **Programmable RF ASICs**



 Saphyrion (former Nemerix) RF ASIC chipset developed under ESA R&D programmes and used for POD applications in Swarm, EarthCare and Sentinel 1a/b, 2a/b, 3a/b.



- · RF performance is even more important for Radio Occultation (carrier phase measurements)
  - Good filtering against interference (Search & Rescue payloads)
  - Good frequency plan (e.g. integer values between all clocks in the receiver)
  - Clock coherency between bands
  - Low phase noise at 1 Hz & short term stability
- ESA is preparing the next generation of RF ASICs:
  - For POD, Radio Occultation and GNSS-R receivers
    - good RF performance and miniaturisation is important
  - Compatible with AGGA-4 (including 3-bit Intermediate Frequency sampling)

## EQM model (example with AGGA-2)





# New GNSS signals and constellations (Galileo, modernized GPS, others)



#### More **robustness** thanks to:

- More signals in Open Service => (error detection / correction)
- no semi-codeless: dual frequency available also with low SNR
- <u>Pilot components</u> (no bit wiping) => very good for EO needing carrier measurements.
- secondary codes: 'lengthen' the spreading code, better autocorrelations while fast acquisition

#### **Small improvement in accuracy** (signals with better codes, but similar carrier)

- Similar signal power levels
- higher code bandwidths (e.g. 10 MHz), BOC modulations
- but similar carrier measurements (driver in EO applications)

#### The new GNSS signals imply:

- · Components more flexible and with more digital processing
  - more channels to improve robustness and RO measurements
  - more digital functions (e.g. digital down conversion, carrier & code aiding, etc).
  - Flexibility (e.g. LFSR & memory-based code generators, more frequency plans)
- Different software: no codeless processing or bit wiping, but more available signals

**ESA preparing** the AGGA-4, RF ASICs and antenna <u>components</u> compatible with new GNSS. We can <u>start</u> developing the <u>receivers</u> (ASIC final pin layout known soon)

#### Conclusion



Applications:

- cm accuracy in Precise Orbit Determination demonstrated (GOCE)
- Radio Occultation: excellent performance of MetOp GRAS

AGGA-2 baseband processors: widely used in ESA and non-ESA missions

ESA preparing the next generation of key GNSS receiver components that can be used for both POD and RO

- AGGA-4 : compatible with Galileo, modernized GPS, Glonass, Beidou.
  Higher number of channels. Expected ASIC samples: 4Q-2011 for all European space industry under equal basis.
- RF ASIC also important in performance and miniaturisation

We start developing the receivers compatible with future signals:

- more robustness (e.g. higher number of Open Signals, pilot components),
- little accuracy improvements