

USE OF COTS PROCESSORS IN SPACE

ADCSS09, ESTEC

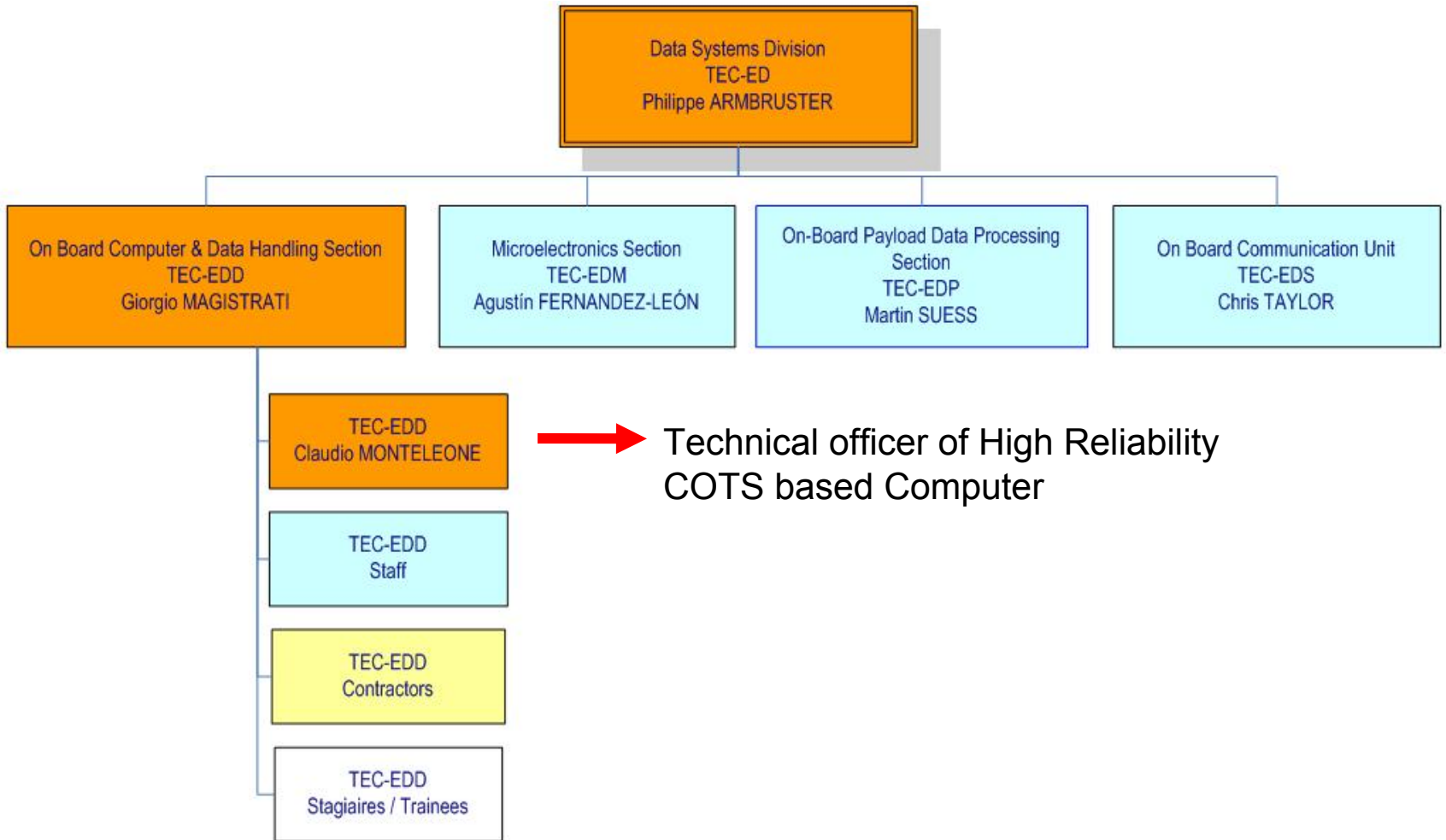
MPSA - Micro-Processors for Space Applications round table

C. Monteleone

On Board Computer & Data Handling Section (TEC-EDD)

ESA ESTEC

TEC-ED Sections



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What we mean with COTS components

A common definition for COTS:

Commercial off-the-shelf (COTS) is a term identifying software or hardware products that are ready-made and available for sale or license

- Manufacturer's standard products
- Can be found in data books
- Usually fast procurement

There are COTS products also for Rad-Hard processors e.g. TSC695, AT697, AT791

Additional definition for COTS components for space:

- Components with no pre-requisite specification respect to space environment (thermal, mechanical, radiation aspects)
- Lower cost respect to similar rad-hard components

Concerns using COTS processors

- ***Radiation effects***
 - Availability of Radiation Tests results are often missing
- ***Variability***
 - A COTS manufacturers do not control the technology parameters that condition the radiation hardness
 - Process is likely to be modified at anytime, tracing the origin and manufacturing process is difficult
 - Processor memories COTS devices has a variability from one manufacturer to another and for a single manufacturer
- ***Obsolescence***
 - COTS suppliers constantly introduce new products while
 - hardened OBC have a long development time and a long life cycle
- ***Power consumption***
 - Higher power consumption than High Rel devices
 - Power consumption stability vs aging effects
- ***Reliability***
 - Reliability data often missing or incomplete
- ***No access to the intimate design***
 - Difficulty to fully characterize the design and to develop models

Motivations for using COTS processors

- High computing performance
- Reduced procurement cost
- Large availability of support tools for SW development and tests
- Large availability of existing software libraries
- Compatibility with ground processors allows developing of low cost test environments and simulators
- Many COTS developed processors technology using sub-100 nm silicon techniques include internal error detection and correction features that can make easier the implementation in computers for space applications of architectural solutions for FDIR.

On-going ESA activity related to COTS processors in space

Title:

COTS based Computer for On board Systems (CoCs)

Main objective:

Study and design for on-board computing systems based on “Commercial Off The-Shelf” components

Activity phases:

1. The design phase

Defining the COTS computers as well as the methods for their manufacturing and qualification

2. The implementation and qualification phase

Manufacturing of EM/EQM breadboards that target real missions.

3. The transfer phase

Bridging the gap between R&D and operational use within projects.

CoCs Activity design phase

Main objectives:

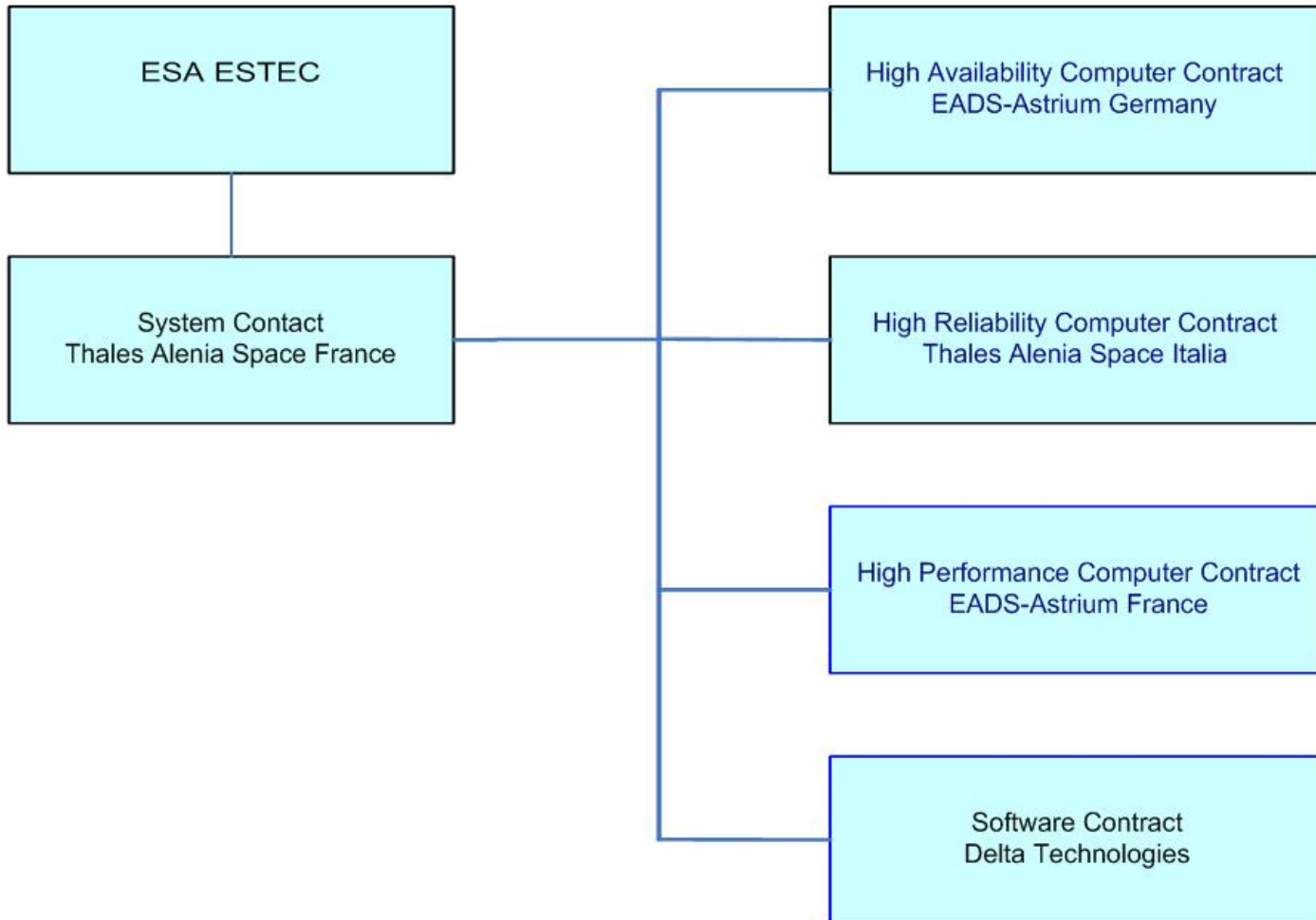
- The selection of the COTS components
- The detailed specification
- Models for the CoCs and its building blocks.
- Methods for the prediction and evaluation of the performances.
- The specification of the CoCs Software

Goal is prototyping 3 COTS based computer architectures:

- The Highly Reliable CoCs (Hi-R)
- The Highly Available CoCs (Hi-V)
- The High Computing Power Payload CoCs (Hi-P)



CoCs Activity Contracts



Hi-R Computer requirements

Highly reliable computers are used to run functions over a quite long period of time while they can be interrupted from time to time without putting the mission in danger.

- Lifetime: 15 years
- A permanent failure: must not lead to mission loss.
- A survival mode is used as ultimate barrier ensuring satellite safety.
- Tolerated outage: 10 seconds maximum.
- Rate of outage: one outage every 30 days.
- Reliability > 0.95 over 15 years.
- Performance: > 200 MIPS.
- Communication services shall be able to manage at the same time up to:
 - 3 high speed bus (> 300 Mbit/s)
 - 3 low speed bus (of the class 1 Mbit/s)
 - 100 low speed I/O's (few Kbit/s)

Hi-V Computer requirements

Highly available computer shall provide dependable services never interrupted during a limited period of availability

- Lifetime: 15 years.
- A permanent failure must not lead to mission loss.
- Duration of the availability period: 30 consecutive days.
- No outage is allowed during the availability period.
- No survival mode is possible.
- Probability of failure during availability period is 10^{-7} per hour.
- Number of availability periods during lifetime: 50.
- Reliability outside the availability period > 0.95 over 15 years.
- Performance: > 200 MIPS.
- Communication services shall be able to manage at the same time up to:
 - 3 high speed bus (> 300 Mbit/s)
 - 3 low speed bus (of the class 1 Mbit/s)

Hi-P Computer requirements

High Processing Power Payload Computer performance figures shall be the following:

- Lifetime: 15 years.
- Duration of availability period: 10 days.
- Outage allowed during availability period: 10 second.
- Number of outage allowed during availability period: 5.
- Failed computer must go automatically in a safe state.
- Number of availability periods during lifetime: 100.
- Performance: > 500 MIPS or > 500 MFLOPS.
- Communication services shall be able to manage at the same time up to:
 - 3 high speed bus (> 300 Mbit/s)
 - 3 low speed bus (of the class 1 Mbit/s)

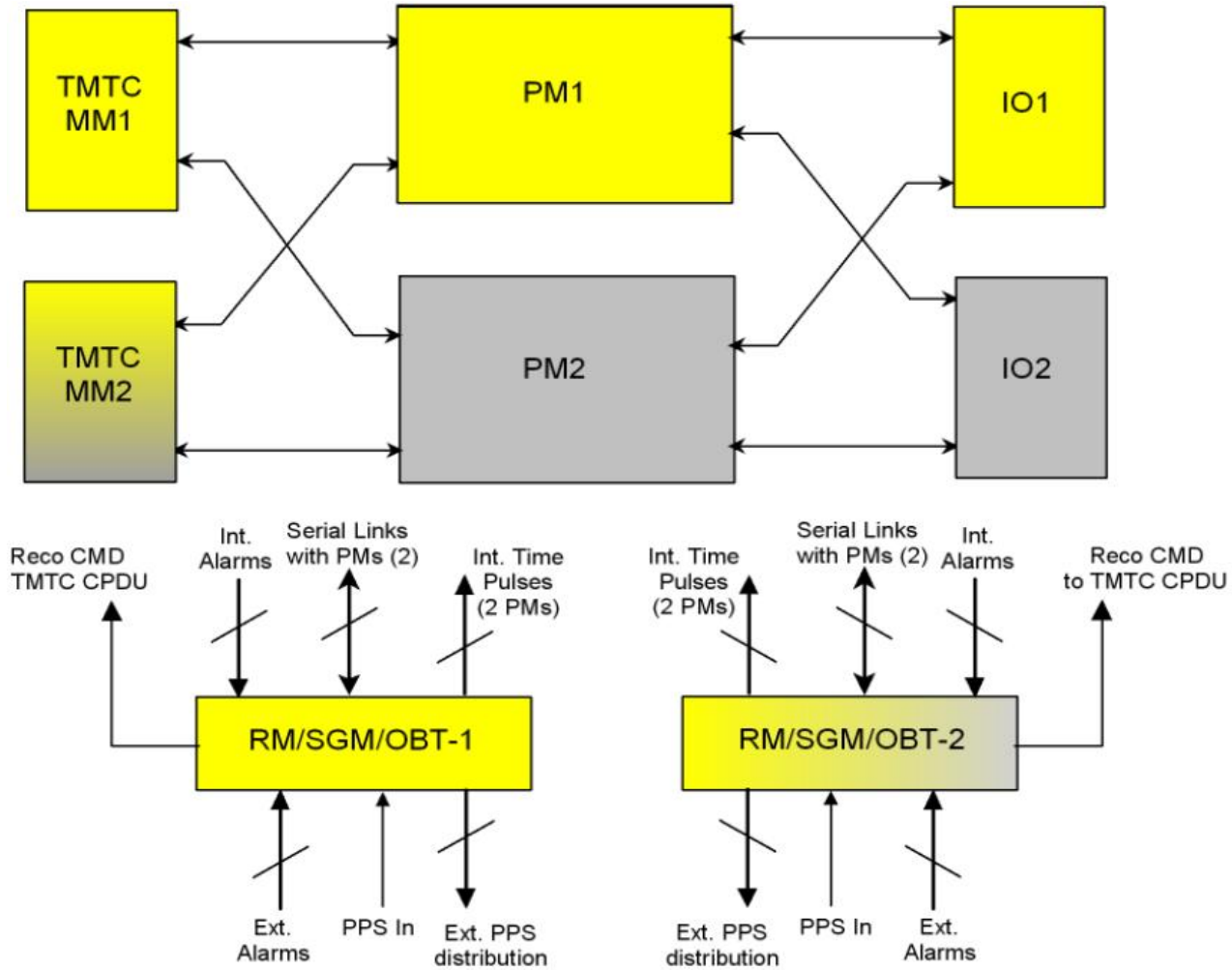
COTS processor selection criteria

- Maturity and Stability of the design and manufacturing process
- Rapid Obsolescence / long term availability
- Diffusion in commercial and embedded markets
- Availability and maturity of development tools and SW
- Availability of up-screened versions
- Access to manufacturers data
- No ITAR or other export restrictions
- Use of silicon technologies having intrinsic radiation tolerance capabilities and availability of Radiation tests results
- Internal error detection and correction features
- Power consumption
- Packaging

COTS Processor selected

Features	PPC7448 (Freescale)	AT697 (ATMEL)
Clock frequency	1.3 Ghz	100 MHz
Dhrystone 2.1 MIPS	3000	86
Manufacturing process	90 nm CMOS SOI	180 nm ATC18RHA
On-chip L1 cache	32 Kbyte Icache with parity 32 Kbyte Dcache with parity	32 Kbyte Icache with parity 16 Kbyte Dcache with parity
On-chip L2 cache	1 Mbyte with tag parity and ECC on data	-
Core	e600 core (PowerPC G4)	LEON2-FT (Sparc V8)
Package	CLGA 360	MCGA 349
SEU (cm ² /device)	3.0x10 ⁻⁴	1.0x10 ⁻⁵
Power (mW/MHz)	9.2	10

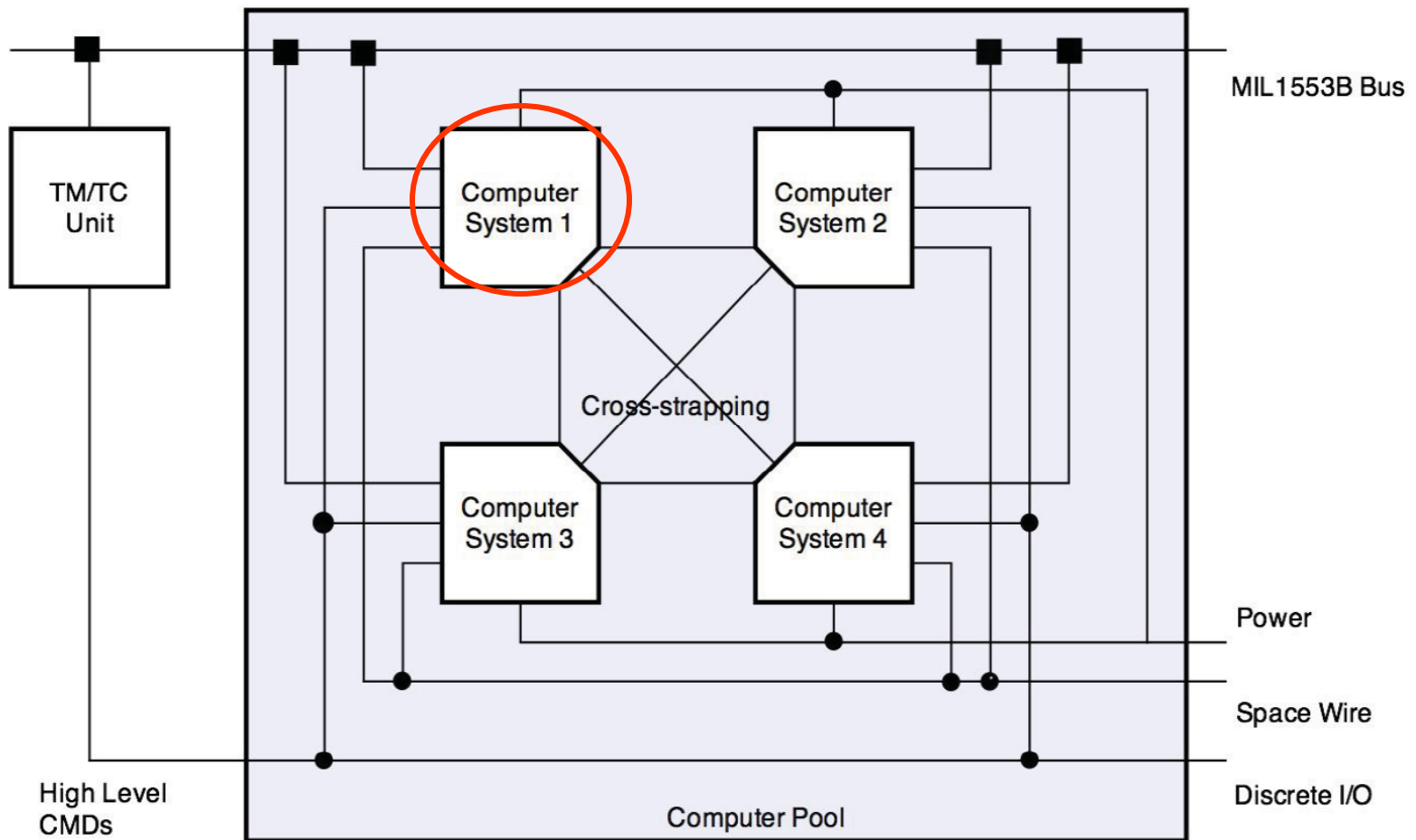
Hi-R Computer Architecture



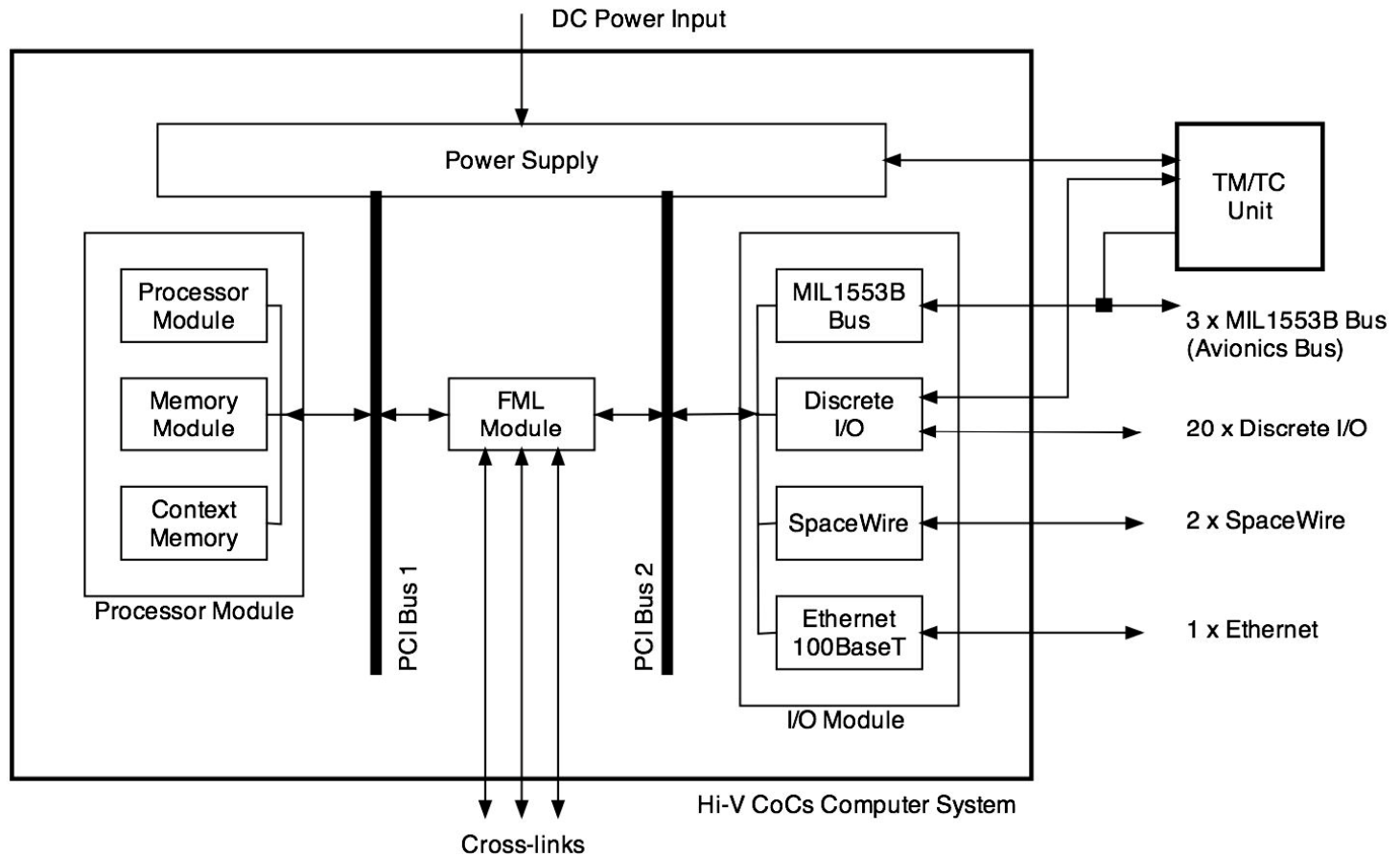
Hi-R FDIR strategy

- Approach is to maintain to a large extent the architecture, redundancy concept and partitioning of functions in use within typical Platform Control computers.
- All the countermeasures needed to minimize occurrence and to solve transients faults are implemented internally to each module
- In particular in the Processor Module this is achieved by a combination of hardware and software features:
 - Volatile memory protected by ECC code
 - Each volatile memory device powered independently to recover from SEFI
 - Protected volatile memory areas
 - Smart watch-dogs
 - HW scrubbing
 - Time redundancy: tasks are computed twice
- The Reconfiguration Module intervenes only in case of locally unrecoverable situations.

Hi-V Computer Architecture



Hi-V Computer System



Hi-P COTS based Computer

The architecture is based on two main modules: Processor module and Smart-I/O module

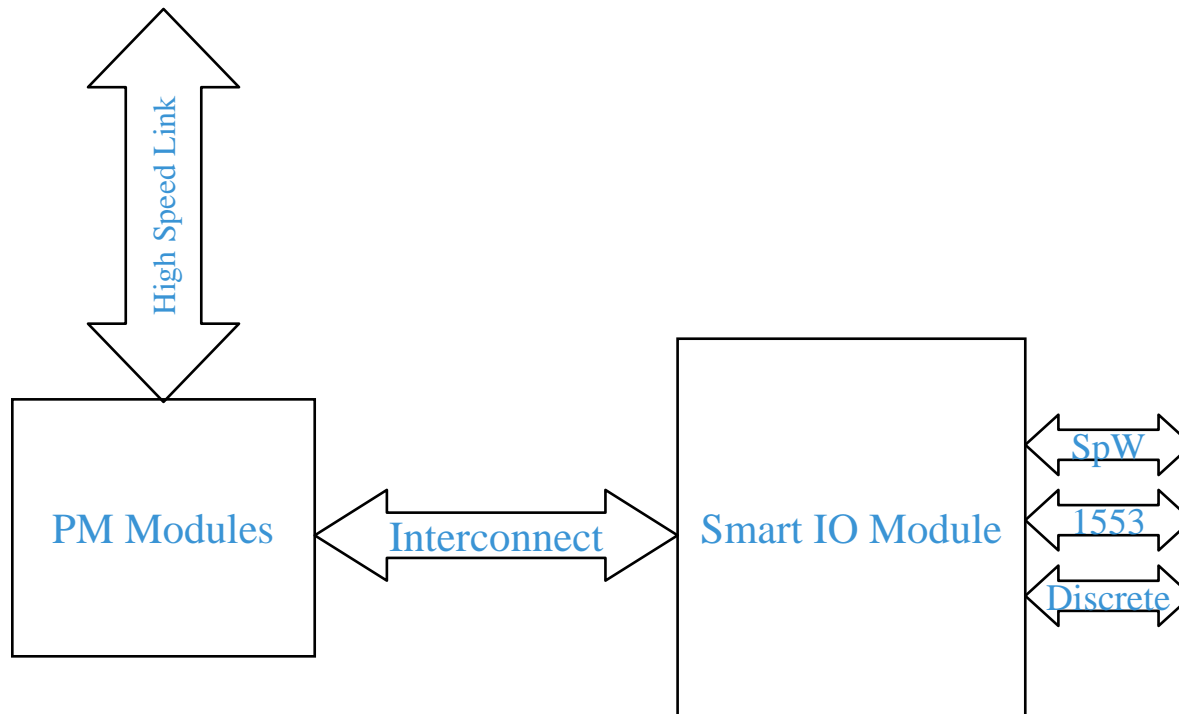
Processor Module:

- It is based on PowerPC-7448
- In case of specific higher performance or lower power consumption needs a DSP based design is used

Smart I/O module:

- It is SEU immune and based on a Rad-Hard but lower performance processor (e.g. AT697)
- Can control multiple Processor Modules
- It is in charge to ensures safe instrument control and to format the data before the distribution in the system
- Can perform low performance phases when PM boards are off.

Hi-P System Interfaces



PM Module - Smart I/O Interconnection candidates:

- cPCI@33MHz allows 1 Gb/s data rate
- SpaceFiber allows 1- 10 Gb/s data rate
- The Wizard Link (Texas Instrument)